## Beam Analysis by the Direct Stiffness Method Steven Vukazich

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## Stiffness Based Approach

Can think of a structure acting as a multi-dimensional spring


## Beam Element Stiffness Matrix in Local Coordinates

Consider an inclined beam member with a moment of inertia $I$ and modulus of elasticity $E$ subjected to shear force and bending moment at its ends.
$i=$ initial end of element
$j=$ terminal end element
$x$ axis (local 1 axis in SAP 2000)


Note the sign convention

We want to find this $4 \times 4$ matrix

$$
\{Q\}=[k]\{\delta\}
$$

Each Column of the Frame Element Matrix in Local Coordinates is derived from Indeterminate Fixed End Moment Solutions


Find the First Column of the Frame Element Stiffness Matrix in Local Coordinates

$$
\begin{aligned}
& \text { set } \Delta_{i}=1 \\
& \text { and } \theta_{i}=\theta_{i}=\Delta_{j}=0
\end{aligned}
$$

## Frame Element Stiffness Matrix

$$
\begin{aligned}
& \left\{\begin{array}{l}
V_{i} \\
M_{i} \\
V_{j} \\
M_{j}
\end{array}\right\}=\left[\begin{array}{cccc}
\frac{12 E I}{L^{3}} & \frac{6 E I}{L^{2}} & -\frac{12 E I}{L^{3}} & \frac{6 E I}{L^{2}} \\
\frac{6 E I}{L^{2}} & \frac{4 E I}{L} & -\frac{6 E I}{L^{2}} & \frac{2 E I}{L} \\
-\frac{12 E I}{L^{3}} \\
\frac{6 E I}{L^{2}} & -\frac{6 E I}{L^{2}} & \frac{12 E I}{L^{3}} & -\frac{6 E I}{L^{2}} \\
\frac{2 E I}{L} & -\frac{6 E I}{L^{2}} & \frac{4 E I}{L}
\end{array}\right]\left\{\begin{array}{c}
\Delta_{i} \\
\theta_{i} \\
\Delta_{j} \\
\theta_{j}
\end{array}\right\} \\
& \{Q\}=[k]\{\delta\}
\end{aligned}
$$

## Structure Stiffness Matrix

Consider a beam comprised of two elements
Each beam joint can move in two directions: 2


The $6 \times 6$ structure stiffness matrix can be assembled

$$
\{F\}=[K]\{\Delta\}
$$ from the element stiffness matrices

## Assemble Structure Stiffness Matrix

1. Number each

2. Connectivity table to assemble structure stiffness matrix

|  | $\mathbf{1}$ | $\mathbf{2}$ | $\mathbf{3}$ | 4 |
| :--- | :---: | :---: | :---: | :---: |
| Associated global DOF for element 1 | $\mathbf{1}$ | $\mathbf{2}$ | $\mathbf{3}$ | 4 |
| Associated global DOF for element 2 | $\mathbf{3}$ | 4 | $\mathbf{5}$ | $\mathbf{6}$ |

## Assembly of Structure Stiffness Matrix from Element Contributions

Form $4 \times 4$ element stiffness matrix for element 1 from $\mathrm{EI}_{1}$ and $\mathrm{L}_{1}$

Form $4 \times 4$ element stiffness matrix for element 2 from $\mathrm{EI}_{2}$ and $\mathrm{L}_{2}$

$$
[k]^{2}=\begin{array}{c|cc:cc|}
\hline 3 & 3 & 4 & 5 & 6 \\
4 & k_{11}^{2} & k_{12}^{2} & k_{13}^{2} & k_{14}^{2} \\
\hdashline 5 & k_{22}^{2} & k_{23}^{2} & k_{24}^{2} \\
\hdashline 5 & k_{31}^{2} & k_{32}^{2} & k_{33}^{2} & k_{34}^{2} \\
& 6 & k_{41}^{2} & k_{42}^{2} & k_{43}^{2}
\end{array} k_{44}^{2},
$$

Label the rows and columns of each $4 \times 4$ element stiffness matrix with corresponding structure DOF from connectivity table

Assemble 6x6 structure stiffness matrix

$[K]=$|  | 1 | 2 | 3 | 4 | 5 | 6 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | $k_{11}^{1}$ | $k_{12}^{1}$ | $k_{13}^{1}$ | $k_{14}^{1}$ | 0 | 0 |
| 2 | $k_{21}^{1}$ | $k_{22}^{1}$ | $k_{23}^{1}$ | $k_{24}^{1}$ | 0 | 0 |
| 3 | $k_{31}^{1}$ | $k_{32}^{1}$ | $k_{33}^{1}+k_{11}^{2}$ | $k_{34}^{1}+k_{12}^{2}$ | $k_{13}^{2}$ | $k_{14}^{2}$ |
| 4 | $k_{41}^{1}$ | $k_{42}^{1}$ | $k_{43}^{1}+k_{21}^{2}$ | $k_{44}^{1}+k_{22}^{2}$ | $k_{23}^{2}$ | $k_{24}^{2}$ |
| 5 | 0 | 0 | $k_{31}^{2}$ | $k_{32}^{2}$ | $k_{33}^{2}$ | $k_{34}^{2}$ |
| 6 | 0 | 0 | $k_{41}^{2}$ | $k_{42}^{2}$ | $k_{43}^{2}$ | $k_{44}^{2}$ |

## Structure System of Equations: Free DOF

## DOF 3 are 4 are free DOF;



DOF 1, 2, 5, and 6 are restrained (support) DOF

$$
\left\{\begin{array}{c}
V_{1} \\
M_{2} \\
V_{3} \\
M_{4} \\
V_{5} \\
M_{6}
\end{array}\right\}=\left[\begin{array}{llllll}
K_{11} & K_{12} & K_{13} & K_{14} & K_{15}{ }^{\prime} & K_{16} \\
K_{21} & K_{22} & K_{23} & K_{24} & K_{25} & K_{26} \\
K_{31} & K_{32} & K_{33} & K_{34} & K_{35} & K_{36} \\
K_{41} & K_{42} & K_{43} & K_{44} & K_{45} & K_{46} \\
K_{51} & K_{52} & K_{53} & K_{54} & K_{55} & K_{56} \\
K_{61} & K_{62} & K_{63} & K_{64} & K_{65} & K_{66}
\end{array}\right]\left\{\begin{array}{c}
\Delta_{1} \\
\theta_{2} \\
\Delta_{3} \\
\theta_{4} \\
\Delta_{5} \\
\theta_{6}
\end{array}\right\}
$$

At free DOF (blue), we know the forces (applied joint loads) but the displacements are unknown

At restrained DOF, we know the displacements but the forces (support reactions) are unknown

## Free DOF System of Equations

Suppose we have loads applied to joint 2
DOF 3 are 4 are free DOF;
DOF 1, 2, 5, and 6 are restrained (support) DOF

$$
\left\{\begin{array}{c}
V_{1} \\
M_{2} \\
V_{3} \\
M_{4} \\
V_{5} \\
M_{6}
\end{array}\right\}=\left[\begin{array}{llllll}
K_{11} & K_{12} & K_{13} & K_{14} & K_{15} & K_{16} \\
K_{21} & K_{22} & K_{23} & K_{24} & K_{25} & K_{26} \\
K_{31} & K_{32} & K_{33} & K_{34} & K_{35} & K_{36} \\
K_{41} & K_{42} & K_{43} & K_{44} & K_{45} & K_{46} \\
K_{51} & K_{52} & K_{53} & K_{54} & K_{55} & K_{56} \\
K_{61} & K_{62} & K_{63} & K_{64} & K_{65} & K_{66}
\end{array}\right]\left\{\begin{array}{c}
0 \\
0 \\
\Delta_{3} \\
\theta_{4} \\
0 \\
0
\end{array}\right\}
$$

At free DOF (blue), we know the forces (applied joint loads) but the displacements are unknown

At restrained DOF, we know the displacements (all equal to zero) but the forces (support reactions) are unknown

## Solve for Displacements at Free DOF



Free DOF Equation Set

$$
\left[\begin{array}{ll}
K_{33} & K_{34} \\
K_{43} & K_{44}
\end{array}\right]\left\{\begin{array}{c}
\Delta_{3} \\
\theta_{4}
\end{array}\right\}=\left\{\begin{array}{l}
V_{3} \\
M_{4}
\end{array}\right\}
$$

Solve for $\Delta_{3}$ and $\theta_{4}$

## Find Support Reactions

DOF 1, 2, 5 and 6 are restrained (support) DOF

$$
\left\{\begin{array}{c}
V_{1} \\
M_{2} \\
V_{3} \\
M_{4} \\
V_{5} \\
M_{6}
\end{array}\right\}=\left[\begin{array}{llllll}
K_{11} & K_{12} & K_{13} & K_{14} & K_{15} & K_{16} \\
K_{21} & K_{22} & K_{23} & K_{24} & K_{25} & K_{26} \\
K_{31} & K_{32} & K_{33} & K_{34} & K_{35} & K_{36} \\
K_{41} & K_{42} & K_{43} & K_{44} & K_{45} & K_{46} \\
K_{51} & K_{52} & K_{53} & K_{54} & K_{55} & K_{56} \\
K_{61} & K_{62} & K_{63} & K_{64} & K_{65} & K_{66}
\end{array}\right]\left\{\begin{array}{c}
0 \\
0 \\
\Delta_{3} \\
\theta_{4} \\
0 \\
0
\end{array}\right\}
$$

After displacements are found, multiply to find unknown forces (support reactions)
$\Delta_{3}$ and $\theta_{4}$, found
from previous step
$\left\{\begin{array}{c}V_{1} \\ M_{2} \\ V_{5} \\ M_{6}\end{array}\right\}=\left[\begin{array}{ll}K_{13} & K_{14} \\ K_{23} & K_{24} \\ K_{53} & K_{54} \\ K_{63} & K_{64}\end{array}\right]\left\{\begin{array}{c}\Delta_{3} \\ \theta_{4}\end{array}\right\}$

